Preliminary analysis on generated jet pressure from ionic polymer composite actuated radial contraction using balloon manometry technique

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Contractile thruster has become a trend for underwater robot propulsion because of its prospective and advantages. However, it is difficult to determine the performance of the generated jet pressure from the small thruster. Thus, in this research an analysis was carried out to observe the radial contraction effects on the jet pressure by applying balloon-manometry technique. Four Ionic Polymer-Metal Composite (IPMC) filament actuators had been utilized as the longitudinal artificial muscle. The lengths of the IPMC actuators were 30 mm, 40 mm and 50 mm. During actuation, the radial contraction differentiated the pressure in the cylindrical shape balloon which was measured by a gauge pressure transducer. The result shows that at 30 mm length, the obtained pressure was almost 0.01psi. The shorter IPMC actuator gave higher contraction force and propulsive pressure.

[Keywords: Jet pressure, Radial contraction, IPMC, Manometry]

Introduction

Currently, rotary-blade propeller is widely used as a thruster for underwater robot and vehicle. However, at small scale specification, the blade propeller has maneuverability and propulsion efficiency problem. Recent development on contractile thruster for underwater robot propulsion had been an alternative method to rotary-blade propeller. It has several advantages over rotary-blade propeller especially for small scale underwater robot such as lower power consumption, provide better maneuverability for robot and less mechanical complexity which leads to maintenance problem such as blade damage and brush motor problem. These advantages are essential criteria for a small underwater robot which has observation and exploring task in robust, high turbidity and complex structure environment such as shipwreck and underwater structure. Inspired by most of the aquatic creatures such as squid, jellyfish and nautilus, this contractile thruster is technically applies water-jet propulsion mechanism. The introduction of deformable smart materials such as shape memory alloy (SMA), piezo material, ionic polymer-metal composite (IPMC) and dielectric elastomer (DE) ensures the designed thruster has near morphological form to the real aquatic animals, whereby those materials had been utilized as active artificial muscle. For instance, Wang et al. had developed SMA wire actuated artificial squid mantle, Shi et al. had developed butterfly inspired thruster and Yeom et al. had utilized IPMC as actuator to mimic jellyfish umbrella. The capability to mimic near morphological design of real system would increase the performance and efficiency of the designed robot. However, it is difficult to determine the performance of the generated jet pressure from the small thruster. Hence, this paper discusses an analysis of the radial contraction actuated by longitudinal IPMC muscles on its resulting jet propulsion pressure under at 3V.

Materials and Methods

In this research, a cylindrical contractile thruster was developed. Four IPMC actuators act as longitudinal muscle for this contractile thruster. Balloon-manometry technique was applied to measure the jet pressure. Observation was made on different actuator’s length and the influence of the actuating pressure on the generated thrust.

Theory

Beam shape IPMC filament bends when opposite polarity with an amount of electrical energy between
1V and 4V was attached at both IPMC surfaces. Tip of the filament bends to positive side polarity as the consequence of free metal-water particle motion in the IPMC. Bending force is determined by the supplied voltage, frequency, type of plated noble metal and thickness of the IPMC. This characteristic allows IPMC filaments works as longitudinal muscle for the cylindrical contractile thruster. As shown in Figure 1, the IPMC filaments were attached to a holding structure in concentricity. Upper IPMC filament bent downward and the bottom IPMC filament bent upward during actuation. On the other hand, the left IPMC filament bent to the right and the right IPMC filament bent to the left. This four inward direction bending motion formed radial contraction and reducing the internal cylindrical volume of the thruster. Bending model for every IPMC filaments was based on cantilever beam model, generated from Euler-Bernoulli Theorem (Figure 2). Different position of the affixed IPMC filaments breaks the mechanics of bending actuation into three cases to measure each of the distributed bending force of the IPMC actuators:

\[ \sum F_{RC} = F_{top} + F_{bottom} + F_{right} + F_{left} \]  

Generally, each of the IPMC actuator tip force, \( F_B \) could be determined by Equation (2), where \( E \) is the Young Modulus of the IPMC actuator, \( L \) is the length of the IPMC actuator, \( \delta \) is the displacement of the IPMC tip and \( I \) is the moment inertia of the beam at different location. However, the \( I \) value differs due to the allocation of the IPMC actuator. For \( F_{top} \) and \( F_{bottom} \), the moment inertia could be calculated using Equation (3) and for \( F_{left} \) and \( F_{right} \), the moment inertia is referring to Equation (4).

\[ F_B = \frac{3EI}{L^3} \delta \]  

\[ I_{top, bottom} = \frac{bh^3}{12} \]  

\[ I_{left, right} = \frac{bh^3}{12} \]

Thrust force, \( T_f \) is the ultimate concern in designing a thruster. Generally, it could be defined as mass flowrate or impulse rate of the jetted fluid. Specifically, thrust force is the momentum of the fluid that obeys 2nd and 3rd Newton Law;

\[ T_f = mv_j + (p_j - p_\infty)A_o \]  

where \( m \) is mass of the fluid, \( v_j \) is the velocity of ejected fluid, \( p_j \) is the jet pressure at the nozzle or orifice, \( p_\infty \) is the ambient pressure and \( A_o \) is the nozzle or orifice opening. It was difficult to measure the mass of the air and thus equation (5) becomes:

\[ T_f = \rho Vv_j + (p_j - p_\infty)A_o \]  

where \( \rho \) is the air density and \( V \) is the volume of the ejected fluid. Using the same Equation (6) above, the velocity of the jet, \( v_j \) could be determined if we could define \( T_f \). Figure 3 shows the mechanics of the propulsion of the thruster.

**Methodology**

IPMC filament is a composite of two layers of noble material such as gold and platinum embedded into an ion exchange polymer sheet such as perfluorinated alkenes or styrene/divinylbenzene-based polymers. Its advantages over the smart materials are including low voltage consumption, easy to be molded and fabricated and able to work underwater. In this study, sulfonate perfluorinated alkenes or Nafion 117 was used as the base ion exchange polymer. Platinum metal was chemically electrodeposited into the surface of the Nafion 117. Fabricated IPMC sheet was cut into four IPMC filaments which had 50 mm x 10 mm x 0.45 mm (LxWxT) dimension each. These IPMC filaments functioned as the artificial longitudinal muscle or the actuators for the cylindrical contractile thruster.

In order to perform the characterization of the actuator, a sample was clamped with different polarity at both surface one of its tip using table top vise. The
electrodes were connected to a power supply (GW Instek PSM-3004). Other tip was touched to a load cell (KYOWA, LVS-5GA) at the same level. Data from this load cell was logged, displayed and monitored using National Instruments Labview software. Supplied voltage was 3VDC and tip force readings were taken at 30 mm, 40 mm and 50 mm IPMC length for every position. For oscillating characterization, the sample was attached to a sweep function generator (GFG-8015G) at square wave 3V supply and 1Hz frequency.

Measurement of the radial contraction was based on manometry technique. As a preliminary experiment, this works was conducted in an open air atmospheric environment and thus the \((p_o - p_{\infty})\) is the differential pressure measured by pressure gauge. A 2.5 mm × 2.5 mm Teflon cube with 9 mm thickness was prepared. Four 0.5 mm thick and 10 mm width copper films which were affixed to the Teflon cube as their electrode. All four copper layers were connected to positive polarity of a power supply (GW Instek PSM-3004). Four IPMC actuators with dimension 50 mm × 10 mm × 0.45 mm (L×W×T) were clamped on top of every copper film. Upper sides of the IPMC filaments were connected to the negative power source. On the other part, a constant 20.70 mm diameter cylindrical nitrile balloon was connected to pressure gauge transducer model PX181B-001G5V Omega Engineering Inc. This balloon was filled with air without overwhelming its constant length diameter. Checked valve was used to sustain the entrapped air pressure at certain level. An Arduino Uno Atmega 328 microcontroller board was utilized to convert the analog data from the pressure transducer to the computer. National Instruments Labview software was utilized to log and displayed the data in graphical form. Measurement was carried out for three different IPMC lengths which are 3 cm, 4 cm and 5cm. Every time the length was changed for measurement, the tip of the IPMC actuator was adjusted to the center of the cylindrical balloon to give uniform actuation deformation force concentration. Data analysis was carried out using Minitab software.

**Discussion**

In average, the recorded thickness of the fabricated IPMC filaments was 0.45 mm. Its average, the surface resistance was 13Ω, measured at two points of 5cm length. Relation between the actuating force over different IPMC actuator length was presented in a graph in Figure 4. Obviously the recorded generated force declined at constant as the length of the IPMC filament increased. The constant value is approximately 0.11 gf per mm of IPMC length. At 30 mm length, the bending tip force could reach up to near 5 gf. The recorded tip force at 50 mm length was 1.4 gf. In addition to this result, Table 1 shows the different actuating force at different position and actuating orientation for different length. The negative result from the bottom actuator was because of the load cell detecting in reverse motion. These data are the key reference in designing the cylindrical contractile thruster. Complex shape and fluid hydrodynamic requires the combination of IPMC actuator length and position for performance optimization.
Figure 5 shows the oscillating speed at three different IPMC actuator lengths respectively. Oscillating speed increased in constant parallel to the increment of the actuator’s length. This phenomenon occurred because longer actuator would have wider bending angle and based on beam bending moment rules where longer actuator would have greater moment. Besides, longer actuator contains more water molecules that could increase the internal stress of the IPMC actuator. Figure 6 depicts the relationship between the generated jet pressure from the contracted balloon at different IPMC filament length. Exhibited pressure was exactly the increment of the internal constant air pressure level caused by IPMC filament actuation. At 3cm length, the recorded pressure was 0.0098 psi (67.57 Nm$^{-2}$) and at 5 cm length, the pressure was 0.0003 psi (2.07 Nm$^{-2}$). This experiment was conducted as a preliminary reference for real application in water. In this case, the medium inside the balloon was natural air which has 1.1839 kg/m$^3$ density at 25°C. Though the density value of water and air is obviously differ, there is no issue as the density is only a constant value in the equation. However, there are several other factors that must be taken into account. These factors would be the balloon material, air compressibility and barometric pressure. In every actuating system, the major concern is to transfer the maximum applied force into the work with minimum waste.

During the radial contraction, the tip force was applied to reduce the balloon radius. Elasticity of the balloon material provides force resistance that could reduce the effect of the tip force. Low butadiene (which impart elasticity to the nitrile copolymer) Nitrile balloon was utilized rather than rubber to reduce the antagonist resistance. Air compressibility had bigger impact on the developed system. The entrapped air might have precise reading to the pressure transducer but less accuracy because of the air compressibility. Just like the resistance of the balloon material, the compressibility of the air absorbs the transferred energy from the IPMC actuation. In real application, the effect of energy absorption could be reduced because water has lower compressibility compared to air and the nozzle is open to jetting out the slug. The other consideration was the barometric pressure. Above the sea level, the IPMC actuators are oscillating easily compared to real underwater application. In submerge state, it might be better in contraction force due to depth level but it could be relatively difficult for the IPMC inflate the thruster during drawing in the water. Once the generated jet pressure had been acquired, the thrust force, $T_f$ would be determined using Equation (7). By assuming that the jet was in cylindrical form, thrust force is the product of jet pressure and the jet cross sectional area. The cross sectional area for the jet was 0.00786 m$^2$. Relation between the generated jet pressure and the resulting thrust force was exhibited in Figure 7.

$$T_f = P_f A_n$$ … (7)

The result shows that the thrust force and jet pressure had linear positive relationship. As noticed in previous chapter, the generated jet

<table>
<thead>
<tr>
<th>Actuating Force</th>
<th>Actuator’s length (mm)</th>
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<tr>
<td>$F_{side}$</td>
<td>3.393 2.864 1.402</td>
</tr>
<tr>
<td>$F_{top}$</td>
<td>5.23 4.65 3.86</td>
</tr>
<tr>
<td>$F_{bottom}$</td>
<td>0.814 0.211 -0.28</td>
</tr>
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Table 1—Characterization of the actuating force at different actuator’s length
pressure decreased if the actuator length increased. Therefore, shorter actuator provided better thrust force. Regression analysis using Minitab software shows that the P value of the variables is 0.004. The linear equation for this relation is:

\[ T_f (N) = 0.00282 + 0.00779 \ P_j (Nm^{-2}) \quad \ldots (8) \]

**Conclusion**

Generally, higher tip force actuation gives better radial contraction and thus the jet pressure. In this experiment, the generated pressure was relatively small, which was around 0.0098 psi for 3cm IPMC filament length. It varies due to the length of the IPMC actuator. The jet pressure value is vital to determine the thrust force generated by the contractile thruster system. Thus, this method provides fundamental guideline to measure the actual water jet thrust force. In further studies, it is suggested to observe the effect of IPMC actuator slope and allocation degree rather than current orthogonal positions.

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